



© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED

FOREWORD

FED. SUPPLY CLASS
53GP

This standard sets forth a standard test method to determine the tensile strength of externally and internally threaded fasteners.

	TABLE OF CO	NTENTS
	<u>Paragraph</u>	Sheet
1. 1.1	SCOPE Applicability	
2. 2.1 2.1.1 2.2	REFERENCED DOCUMENTS Government documents Specifications, standards and handbooks Other publications	
3.	DEFINITIONS	
4. 4.1 4.1.2 4.1.3 4.1.3.2 4.1.3.3 4.1.3.3 4.2 4.2.1 4.2.2 4.2.3 4.2.4 4.2.4 4.2.4.1 4.2.4.2 4.2.4.3 4.2.4.3 4.2.5	GENERAL REQUIREMENTS Test apparatus. Testing machine Extension measuring device Test fixtures NAS1069 fixture Angularity evaluation Test specimens General Externally threaded fasteners Internally threaded fasteners Grip length Specified grip range	4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 5 5 9 9 9 9 10 10
5. 5.1 5.1.1 5.1.1.1 5.2 5.3 5.4 5.5 5.6 6.	DETAIL REQUIREMENTS Test procedures Test Setup NAS1069 Test fixture Tension plate fixture Loading rate Load-deflection curves Failing load determination Yield load determination Number of specimens NOTES	10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 10 11 11 11 11 11 11 11 11
6.1	Test report	

AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC 1000 WILSON BLVD. ARLINGTON, VA 22209

PRODUCT AI			
PRC		THE INITIAL RELEASE OF THIS DOCUMENT SUPERSEDES MIL-STD-1312-8A.	
			REVISION
	PROJECTION	NATIONAL AEROSPACE STANDARDS COMMITTEE	2
10-60	PROCUREMENT SPECIFICATION		CLASSIFICATION STANDARD PRACTICE
FORM	NONE	FASTENER TEST METHODS, METHOD 8, TENSILE STRENGTH	NASM1312-8 SHEET 1 OF 14

Copyright Aerospace Industries Association of America Inc. JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIANAS ANCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS \R APPLICATION OR (2) THE EXISTENCE OF OR APPLIC Not for Resale,2011/9/1 16:56:53 GMT IR TRADEMARKS RIGHTS. ISSUE DATE: AUGUST 1997





© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED

FIGURES

FIGURE 1 – TENSION FIXTURES FOR SHORT GRIP FASTENERS	5
FIGURE 2 – TENSION SETUP	7
FIGURE 3 – TEST PLATE	8
FIGURE 4 – TENSION FIXTURE ASSEMBLY	9
FIGURE 5 – SAMPLE LOAD-DEFLECTION CURVE	12
FIGURE 6 – JOHNSON'S TWO-THIRDS APPROXIMATE METHOD FOR	
DETERMINATION OF YIELD STRENGTH.	13

TABLES

TABLE I – DIMENSIONS
TABLE II – TENSION LOAD RATES



FORM 09-01

(2) COMPLETELY REVISED

REVISION 2 NAS1312-8 SHEET 2

Copyright Aerospace Industries Association of America Inc. JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with Alk/NAS ANCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS\R APPLICATION OR (2) THE EXISTENCE OF OR APPLIC Not for Resale.2011/9/1 16:56:53 GMT/R TRADEMARKS RIGHTS.





© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED

1. SCOPE

1.1 Applicability. This method covers a procedure for testing all types of structural fasteners in axial tension at room temperature. This procedure is intended to define the test required to determine the tensile strength of the fastener itself, not the strength of the fastener in any combination of sheet materials, sheet heat treat levels, or sheet thicknesses.

2. REFERENCED DOCUMENTS

- 2.1 Government documents.
- 2.1.1 Specifications, standards and handbooks. Unless otherwise specified, the following specifications, standards and handbooks of the issue listed in the current Department of Defense Index of Specifications and Standards (DoDISS) and the supplement thereto (if applicable), form a part of this standard to the extent specified herein.

MILITARY STANDARDS

MIL STD 45662 Calibration System Requirements

(Copies of specifications, standards, handbooks, drawings and publications required by contractors in connection with specific acquisition functions should be obtained from the contracting activity or as directed by the contracting officer.)

2.2 Other publications. The following document(s) forms a part of this specification to the extent specified herein. The issues of the documents which are indicated as DOD adopted shall be the issue in the current DoDISS and the supplement thereto, if applicable.

ASTM INTERNATIONAL

ASTM E4 Standard Methods of Force Verification of Testing Machines ASTM E83 Tentative Method of Verification and Classification of Extensometers Systems

(Copies can be obtained from ASTM International, 100 Barr Harbor Drive, PO Box C700, West Conshohocken, PA, 19428-2959 USA <u>www.astm.org</u>)

AEROSPACE INDUSTRIES ASSOCIATION (AIA)

NAS1069 Tension Fatigue Test Procedure for Aeronautical Fasteners

(Copies can be obtained from the Aerospace Industries Association, 1000 Wilson Blvd., Suite 1700, Arlington, VA 22209 <u>www.aia-aerospace.org</u>)

INTERNATIONAL ORGANIZATION FOR STANDARDIZATION (ISO)

ISO 10012 Measurement Management Systems - Requirements for Measurement Processes and Measuring Equipment

(Copies can be obtained from the American National Standards Institute, 25 West 43rd Street, 4th Floor, New York, NY 10036-8002 USA <u>www.ansi.org</u>)

NCSL International (National Conference of Standards Laboratories)

NCSL Z540.3 Calibration Laboratories and Measuring and Test Equipment

(Copies can be obtained from NCSL International, 2995 Wilderness Place, Suite 107, Boulder, CO 80301. <u>www.ncsli.org</u>)

REVISION	
2	
NAS1312-8	1
SHEET 3	

(2) COMPLETELY REVISED

Copyright Aerospace Industries Association of America Inc. 'IPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE 5T7 NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIA/NAS \$NCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS \R APPLICATION OR (2) THE EXISTENCE OF OR APPLIC Not for Resale.2011/9/1 16:56:53 GMTIR TRADEMARKS RIGHTS.

THIS DRAWING SUPERSEDES ALL ANTECEDENT STANDARD DRAWINGS FOR THE SAME PRODUCT AND SHALL BECOME EFFECTIVE NO LATER THAN SIX MONTHS FROM THE LAST REVISION DATE.

=ORM 09-01



© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED



3. DEFINITIONS

Not applicable.

4. GENERAL REQUIREMENTS

- 4.1 Test apparatus.
- 4.1.1 Testing machine. The testing machine shall be capable of applying a tensile load at a controllable rate. The calibration system for the machine shall conform to the requirements of <u>MIL STD 45662</u>ISO 10012 or NCSL Z540.3. Its accuracy shall be verified every 12 months by a method complying with ASTM E4, using a calibration device that shall have been calibrated by the National Institute of Standards and Technology not more than 2 years prior to its use. The yield loads, ultimate loads, and structural failure loads of the fasteners tested (5.4 and 5.5) shall be within the loading range of the testing machine as defined in ASTM E4.
- 4.1.2 Extension measuring device. The extension measuring device shall be an averaging, differential-transformer extensioneter, or equivalent, preferably of the separable type. It shall conform to the requirements of ASTM E83, Class B-1 when used in conjunction with an autographic recorder. The extensioneter shall be capable of installation so as to measure either the relative movement between the movable and stationary crossheads of the testing machine or the extension of the fastener only, preferably the latter. Load and extension ranges shall be used to give the initial portion of the load-extension curve a slope between 45 and 60 degrees.
- 4.1.3 Test fixtures. Test fixtures specified in 4.1.3.1 and 4.1.3.2 are permissible. Other types of fixtures may be used provided all of the following requirements are met:
 - a. Hole size for any fastener shall be the maximum shank diameter of the fastener to be tested plus 0.001 inch with a tolerance of +0.004, -0.000 inch. Alternatively, the hole shall conform to the diameter and limits specified by the fastener manufacturer or by the applicable procurement specification.
 - b. The hole shall be perpendicular to the fixture surfaces within ±1 degree.
 - c. Where applicable, the hole shall be chamfered to provide clearance for the head-to-shank fillet radius of the fastener. The diameter of the chamfer shall not exceed the maximum diameter of the shank of the fastener being tested, plus two times the maximum fillet radius of the fastener being tested, plus .010 inch. When a chamfer at the entrance of the hole is used, the diameter of the chamfer shall be recorded in the Test Report. See 6.1.
 - d. The fixture shall be capable of applying an axial tensile load through the center line of the fastener by means of suitable supporting fixtures.
- 4.1.3.1 NAS1069 fixture. This recommended test fixture is either the cup or insert type as defined in NAS1069, figures 8-13. Hole size for any unspecified fastener shall be the maximum shank diameter of the fastener to be tested plus 0.001 inch with a +0.004, -0.000 inch tolerance. In any case, the hole in the fixture shall not reduce the bearing area of the fastener head by more than 10 percent. The bearing area shall be considered as the area between the minimum bearing face outer diameter and the maximum diameter at the point of tangency of the head-to-shank fillet radius. The fixture design shall provide clearance for the head to shank fillet radius.
- 4.1.3.2 Tension plate fixture. The tension plate type fixture may be used for short grip length fasteners. The fixture configuration shall be as shown in Figures 1 and 2 or Figures 3 and 4. The upper and lower plates shall be parallel within 0.25 degree.

REVISION 2
51312-8
SHEET 4

Copyright Aerospace Industries Association of America Inc. 'JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIANAS NCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS \R APPLICATION OR (2) THE EXISTENCE OF OR APPLI Not for Resale,2011/9/1 16:56:53 GMTIR TRADEMARKS RIGHTS.

FORM 09-01

(2)

COMPLETELY REVISED

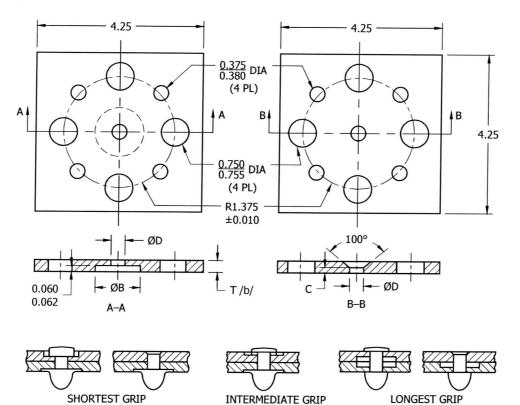




© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED

4.1.3.3 Angularity evaluation. When tests are intended to evaluate the effect of angularity under the head or nut (collar) end of the fastener, the angularity shall be applied by the use of tapered spacers with the required angularity. The spacers shall be made of material harder than that of the fastener and shall have flat ground surfaces. The area of the spacer shall be greater than the bearing area of the fastener head or the nut being tested. The hole shall be the maximum shank diameter plus 0.005 inch with a tolerance of +0.003, -0.000 inch. Alternatively, the hole size shall conform to the diameter and limits specified by the fastener manufacturer or applicable specification.

4.2 Test specimens.



/a/ Omit counterbore and use 2024 T3 or 2024 T4 aluminum sheet when testing DEUTSCH blind rivets, unless otherwise specified.

/a/ /b/ /c/

REVISION 2

NAS1312-8

SHEET 5

/a/ /b/ /c/

- /b/ Counterbore optional, if short and intermediate grips are not required.
- /c/ Surface Roughness 32 to 63 microinches Ra per ASME B46.1.
- (d) Material: 4140 Alloy steel heat treated to 200/220 KSI 200,000 psi, or equal /1/.
- (e) Tolerances unless otherwise specified: decimals ± 0.010 , angels ± 1 degree.
- (f) Chamfer all hole edges to provide head fillet clearance.
- (g) Dimensions in inches.

/a/ /b/ /c/

FIGURE 1 – TENSION FIXTURES FOR SHORT GRIP FASTENERS

0-60	
ORM	(2

2) COMPLETELY REVISED

Copyright Aerospace Industries Association of America Inc. 'JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIA/NAS \NCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS \R APPLICATION OR (2) THE EXISTENCE OF OR APPLIC Not for Resale,2011/9/1 16:56:53 GMT/R TRADEMARKS RIGHTS.





© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED

					TA	BLE I –	DIMENS	SIONS							
Nom	Т	ØВ		Ø	D, +0.00	4, -0.00	0			C di	mensior	+0.0	00 -0	002	
Fstnr	+0.001	±0.01	/a/ /b/	/e/	/f/	/g/	/h/	/j/	/a/	/b/ /h/			/e/	/ f/	/g/
Size	-0.000		/c/ /d/			. 5.		1.31	, -,	/ 5/ / 11/	10/13/	/ 4/	/ / /	177	797
0.093	0.125	0.75	0.094				0.097			0.036					
0.125	0.125	0.75	0.125			0.136	0.129			0.042		0.030			0.039
0.156	0.188	0.88	0.156	0.164	0.170	0.173	0.160			0.055				0.070	
0.164	0.188	0.88	0.165						0.072			0.050	0.072	0.070	0.015
0.190	0.250	1.00	0.190	0.199	0.197	0.204	0.192	0.190	0.084	0.070	0.048	0.046	0 084	0.000	0.064
0.190	0.250	1.00	0.190	0.199	0.197				0.084	0.070	0.048			0.000	
0.250	0.250	1.00	0.250	0.261	0.258		0.256	0.260		0.095		0.061			
0.313	0.250	1.12	0.312	0.313	0.339				0.138	0.106	0.070			0.100	
0.375	0.250	1.12	0.375	0.375	0.391			1 1		0.134	0.081		0.167		

/a/ Protruding and flush tension head solid shank fasteners.

/b/ Flush (MS20426) head solid shank fasteners.

/c/ Flush head solid shank fasteners.

/d/ Flush shear head driven rivets.

/e/ 3O bolts NAS1675 Blind Fasteners and NASM81177 Blind bolts.

/f/ Blind bolts.

/g/ Explosive Rivets.

/h/ Blind rivets except DEUTSCH.

/j/ DEUTSCH blind rivets. /1/

(k) Dimensions in inches.

FORM 09-01

(2) COMPLETELY REVISED

AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC 1000 WILSON BLVD. ARLINGTON, VA 22209

> REVISION 2 NAS1312-8 SHEET 6

Copyright Aerospace Industries Association of America Inc. 'JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIA/NAS \NC: OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS\R APPLICATION OR (2) THE EXISTENCE OF OR APPLIC Not for Resale,2011/9/1 16:56:53 GMT/R TRADEMARKS RIGHTS.



AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC 1000 WILSON BLVD. ARLINGTON, VA 22209

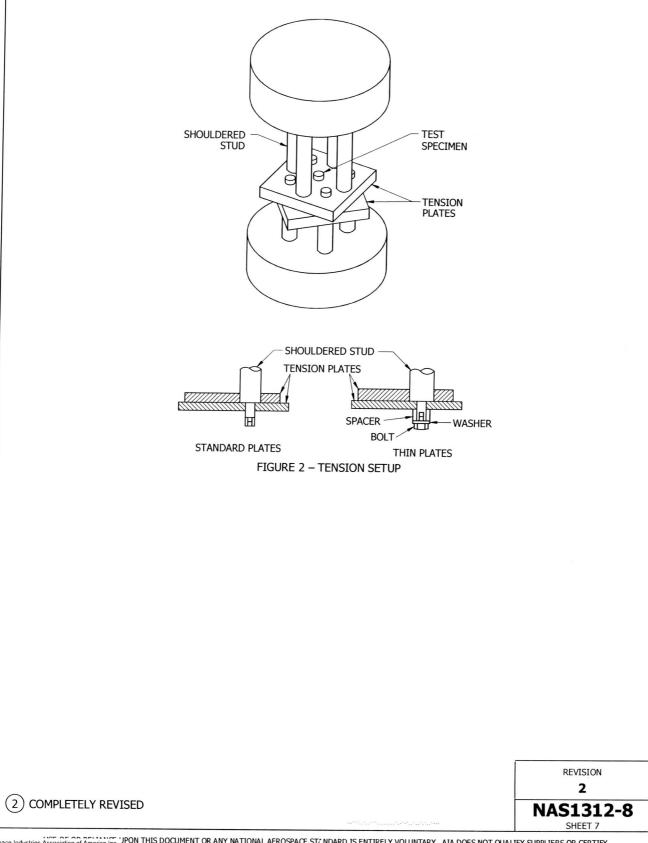
THIS DRAWING SUPERSEDES ALL ANTECEDENT STANDARD DRAWINGS FOR THE SAME PRODUCT AND SHALL BECOME EFFECTIVE NO LATER THAN SIX MONTHS FROM THE LAST REVISION DATE.

FORM 09-01

NATIONAL AEROSPACE STANDARD



© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED



Copyright Aerospace Industries Association of America Inc. 'JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIA/NAS \$NCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS \R APPLICATION OR (2) THE EXISTENCE OF OR APPLIC Not for Resale.2011/9/1 16:56:53 GMT/R TRADEMARKS RIGHTS.

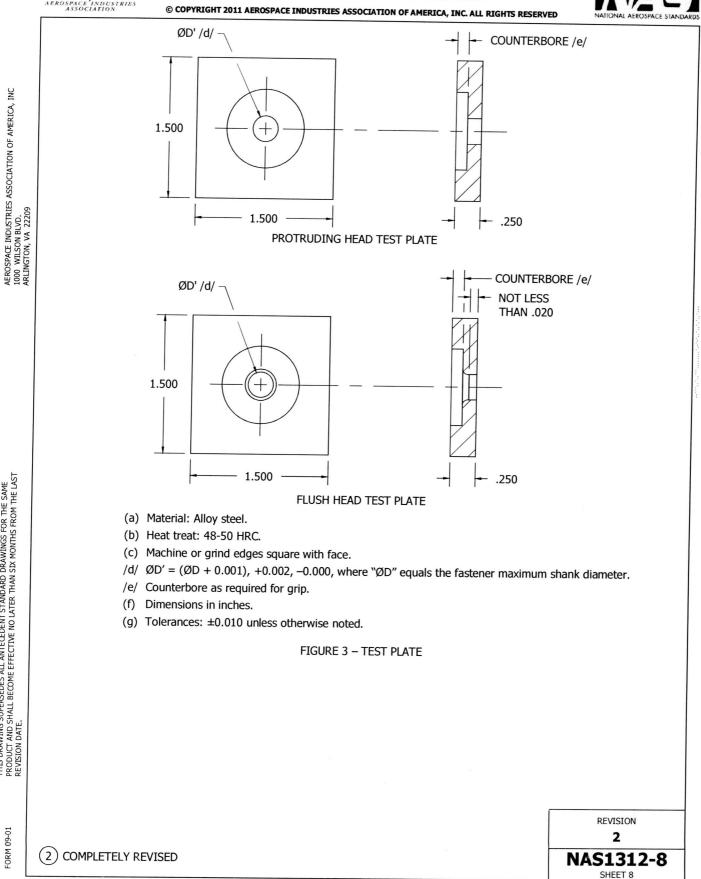


THE DRAWING SUPERSEDES ALL ANTECEDENT STANDARD DRAWINGS FOR THE SAME PRODUCT AND SHALL BECOME EFFECTIVE NO LATER THAN SIX MONTHS FROM THE LAST REVISION DATE.

FORM 09-01

NATIONAL AEROSPACE STANDARD





Copyright Aerospace Industries Association of America Inc. 'JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIANAS ANCE OF TIEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD' Sold to INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS \R APPLICATION OR (2) THE EXISTENCE OF OR APPLIC Not for Resale.2011/9/1 16:56:53 GMT/R TRADEMARKS RIGHTS.



INC

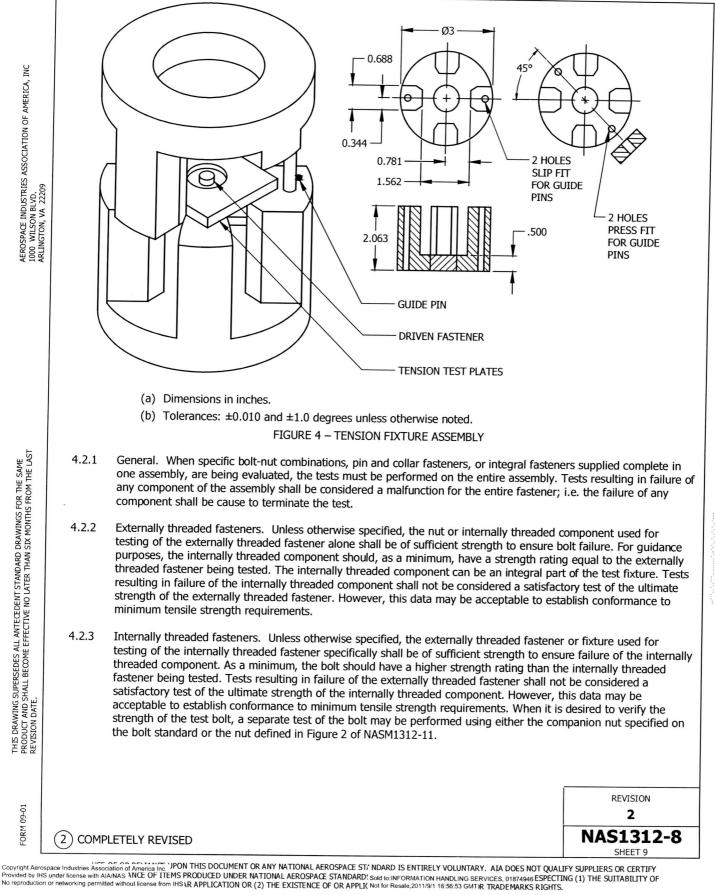
AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA,

FORM 09-01

NATIONAL AEROSPACE STANDARD



© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED







© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED

- 4.2.4 Grip length.
- 4.2.4.1 Specified grip range. Fasteners having a specific grip range (such as lockbolts, HI-LOKS, blind rivets, and blind bolts) shall be tested in such grip conditions as required by applicable specifications or procurement documents.
- 4.2.4.2 Unspecified grip range. Threaded fasteners having no specified grip range, such as those threaded to the head, shall be tested at the grip length where, as a minimum, the chamfer at the end of the bolt threads must project beyond the end of the nut.
- 4.2.4.3 Grip length adjustment. The grip length for fasteners which must be installed with installation preload shall be adjusted with washers or shims located between the test fixtures, not under nuts, collars or heads.
- 4.2.5 Washers or spacers. The use of washers or spacers under fastener heads, nuts or collars is specifically prohibited unless the use of such washers or spacers is specifically required in the product application. Where spacers are required to evaluate the effect of angularity or to provide clearance for the head to shank fillet radius, they shall be hardened and ground as specified in 4.1.3.3.

5. DETAIL REQUIREMENTS

- 5.1 Test procedures.
- 5.1.1 Test Setup.
- 5.1.1.1 NAS1069 Test fixture. The specimen shall be installed in the holding fixture at the proper grip using normal installation procedures. Preload shall not be used except for those fasteners where preload is a normal result of the installation. The assembly of specimen and fixtures shall be placed between the machine tension heads. The difference between the maximum and the average stress caused by misalignment or eccentric loading shall not exceed 3 percent average stress.
- 5.1.1.2 Tension plate fixture. The specimens shall be installed in the fixture at the proper grip using normal installation procedures. Preload shall not be used except for those fasteners where preload is a normal result of the installation. The assembly of specimen and plates shall be installed in a suitable test jig as shown in Figures 2 or 4 and placed between the compression heads of the testing machine. Care shall be exercised to locate the jig at the center of the piston when hydraulic testing machines are used.
- 5.2 Loading rate. Tension load shall be applied to the fastener slowly and evenly. Unless otherwise specified by the fastener manufacturer, or by applicable detail specification or procurement document, the rate of loading shall be in accordance with Table II. Larger or smaller fasteners than those shown in Table II shall be tested at rates producing 100,000 pounds/minute (±10 percent) per square inch of nominal shank area.

THIS DRAWING SUPERSEDES ALL ANTECEDENT STANDARD DRAWINGS FOR THE SAME PRODUCT AND SHALL BECOME EFFECTIVE NO LATER THAN SIX MONTHS FROM THE LAST REVISION DATE.

FORM 09-01

2 COMPLETELY REVISED

REVISION
2
NAS1312-8
SHEET 10

Copyright Aerospace Industries Association of America Inc. 'JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIA/NAS \$NCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS \R APPLICATION OR (2) THE EXISTENCE OF OR APPLIC Not for Resale,2011/9/1 16:56:53 GMT IR TRADEMARKS RIGHTS.





© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED

	TABLE II – TENS	ION LOAD RATES	
Nominal	Load rate	Nominal	Load rate
fastener	(lb./min ± 10%)	fastener	(lb./min ± 10%)
diameter		diameter	
0.125	1,240	0.563	24,800
0.156	1,920	0.625	30,600
0.164	2,100	0.750	44,000
0.190	2,800	0.875	60,000
0.250	5,000	1.000	78,000
0.313	7,700	1.125	100,000
0.375	11,000	1.250	122,000
0.438	15,000	1.375	148,000
0.500	19,600	1.500	176,000

- (a) As an alternative loading method, the rate of loading can be determined as a constant head travel that gives strain rates consistent with the above load rates in the elastic range.
- 5.3 Load-deflection curves. Load-deflection curves, when specified, shall be made by autographic recording. The movement sensing element shall be so installed as to either measure the relative movement between the movable cross-head and the stationary cross-head, or to measure the deflection of the fastener only, preferably the latter. The latter measurement shall be accomplished by placing the movement sensing elements adjacent to the fastener between the facing surfaces of the fastener holding fixtures for the NAS1069 fixture, or on the externally opposed surfaces of the tension plate, or over the ends of the fastener. The fixture shall have the capability of being non-yielding at ultimate load.
- 5.4 Failing load determination. When a structure failure occurs before the ultimate or fracture load has been recorded, it may be detected on the load-deflection curve as a peak load followed by severe permanent deformation without increase in load or with a decrease in load. A sample load-deflection curve illustrating a "structural failure" is shown on Figure 5. The first peak load shall be designated as the "structural failure load" and the second peak shall be the "ultimate load."
- 5.5 Yield load determination. The yield load shall be determined by the Johnson's two-thirds method of approximation, as shown on Figure 6. On the load-deflection plot described in 5.3, draw a line (line "A") with two-thirds the slope of the straight portion of the curve. Parallel to this line and tangent to the load-deflection curve, draw another line (line "B"). The point of tangency of line "B" with the curve represents the yield load.
- 5.6 Number of specimens. Unless otherwise specified in the detail specification or purchase order, a minimum of five (5) specimens shall be tested for each fastener diameter and test condition.

REVISION	
2	
NAS1312-8	
SHEET 11	

(2) COMPLETELY REVISED

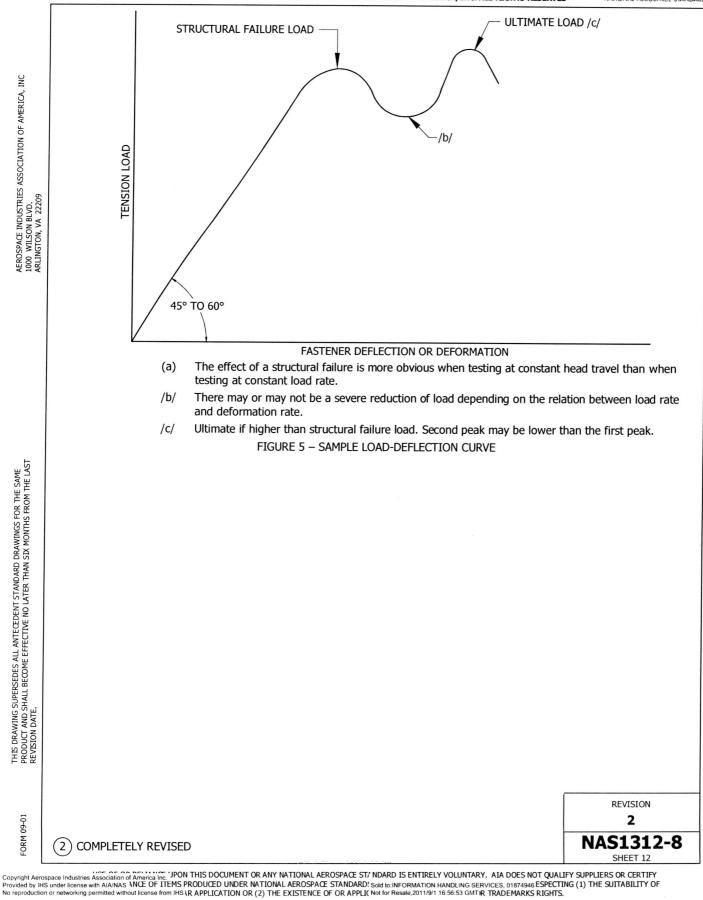
Copyright Aerospace Industries Association of America Inc. 'JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AIA/NAS ANCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD': Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS \R APPLICATION OR (2) THE EXISTENCE OF OR APPLI Not for Resale,2011/9/1 16:56:53 GMT IR TRADEMARKS RIGHTS.

FORM 09-01





© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED



AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC 1000 WILSON BLVD. ARLINGTON, VA 22209

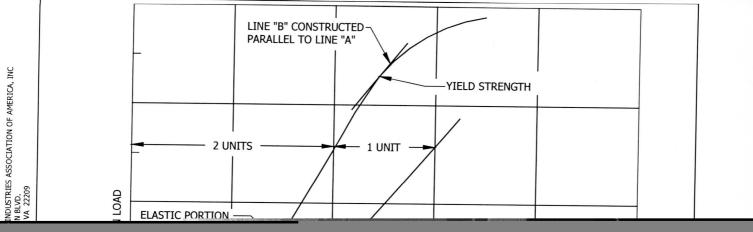
THIS DRAWING SUPERSEDES ALL ANTECEDENT STANDARD DRAWINGS FOR THE SAME PRODUCT AND SHALL BECOME EFFECTIVE NO LATER THAN SIX MONTHS FROM THE LAST REVISION DATE.

FORM 09-01





© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED





AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC 1000 WILSON BLVD. ARLINGTON, VA 22209

NATIONAL AEROSPACE STANDARD



© COPYRIGHT 2011 AEROSPACE INDUSTRIES ASSOCIATION OF AMERICA, INC. ALL RIGHTS RESERVED

- b. Test machine.
 - 1. Model and serial number.
 - Calibration date.
- c. Loading rate or strain rate.
- d. Ultimate loads.
- e. Yield loads.
- f. Installation procedure.
- g. Fixture used.
 - 1. Hole diameter.
 - 2. Chamfer dimensions (for fillet radius clearance).
- h. Structural failure loads.
- i. Mode of failure.
- j. Load-deflection curve.
- k. Strain magnification factor.
- Method of deflection measurement.
- m. Cause and duration of any interruptions during test.
- n. Results of all inspections.

FORM 09-01

(2)

COMPLETELY REVISED

REVISION 2

NAS1312-8 SHEET 14

Copyright Aerospace Industries Association of America Inc. 'JPON THIS DOCUMENT OR ANY NATIONAL AEROSPACE ST/ NDARD IS ENTIRELY VOLUNTARY. AIA DOES NOT QUALIFY SUPPLIERS OR CERTIFY Provided by IHS under license with AlAXIAS \$NCE OF ITEMS PRODUCED UNDER NATIONAL AEROSPACE STANDARD! Sold to:INFORMATION HANDLING SERVICES, 01874946 ESPECTING (1) THE SUITABILITY OF No reproduction or networking permitted without license from IHS\R APPLICATION OR (2) THE EXISTENCE OF OR APPLI (Not for Resale, 2011/9/1 16:56:53 GMT\R TRADEMARKS RIGHTS.